Imperial College London



Wall Resolved Fluid-Structure Interaction Numerical Simulation of a Modern Wind Turbine Blade

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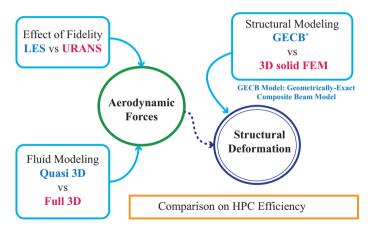
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Objectives







Challenges

Separated flows

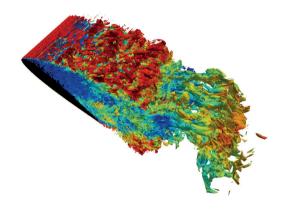
Thick boundary layer Leading/Trailing edges separation Vortex shedding/interactions

 Anisotropic turbulence **High-fidelity methods (LES/DNS)**

High Re flow

Grid Resolution $h/L \propto Re^{-3/4}$ FLOPS scales as Re3

Efficient numerical method Highly scalable



LES simulation of NACA 0012 at $Re = 1.56 \times 10^5$ $\alpha = 16^\circ$ Q-criterion iso-contour colored by velocity





Overview of the two solvers

Nektar++/SHARPy

- High-fidelity (LES/DNS) method
- Quasi-3D approach: Thick-strips
- High-order spectral/hp method
- Beam model for structure
- Open-Source

OpenFoam/CalculiX

- Medium-fidelity (URANS) method
- Full 3D approach
- Second-order Finite-volume
- FEM modelling for structure
- Open-Source

Numerical Method

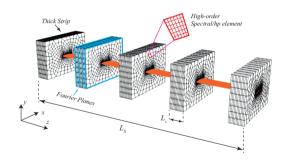




Thick Strip LES method

3D domain models as separate smaller domains: Thick Strips

- Spanwise thickness Lz
- Capturing local 3D turbulence
- Reduces computational expense
- Implicitly connected by structural dynamics
- High order hp elements in xy
- Fourier expansion in spanwise (z) direction



Ref- AIAA SciTech21



Nektar++/SHARPy FSI Solver

Flow Solver

- Transformed Navier-Stokes equations
- Coordinate transformation to avoid remeshing

Inertial Coord. Body-fitted Coord.

$$\bar{\mathbf{X}}(\bar{x},\bar{y},\bar{z}) \to \mathbf{X}(x(t),y(t),z(t))$$

- High-order Spectral/hp element method
- P-order polynomial expansion in xy
- Fourier expansion in z
- High-order stiffy-stable velocity splitting velocity correction scheme
- Nektar++ www.nektar.info

Structural Solver

- Flexible multi-body solver
- Geometrically-Exact Composite Beam model
- Non-linear high deformation
- Static and Dynamic formulation
- Quadratic Finite Element method
- Newmark- β time integration
- SHARPy www.imperial.ac.uk/aeroelastics/sharpy/



OpenFoam/CalculiX FSI Solver

Flow Solver - OpenFoam

- Finite volume method
- OpenFoam solver: pimpleFoam (operating in PISO mode)
- URANS (k-w SST)
- Backward time scheme (Second Order)
- Simulation dt: 5×10^{-5} s
- OpenFoam www.openfoam.com

Structural Solver - CalculiX

- FEM with solid elements
- CalculiX solver: Nonlinear Dynamic Direct solver using Spooles
- Solid Section structure
- Orthotropic Elastic Material
- Simulation dt: 5×10^{-5} s
- CalculiX calculix.de

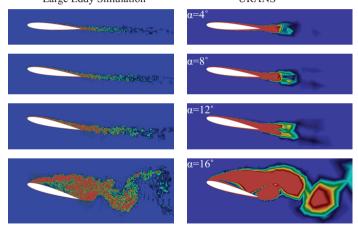
Coupling through preCICE (precice.org)

Flow over NACA0012 section



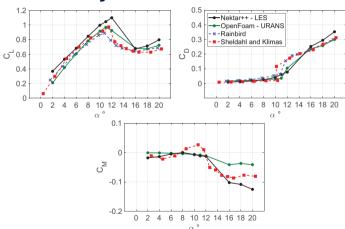


Vorticity contours $Re = 1.56 \times 10^5$ Large Eddy Simulation





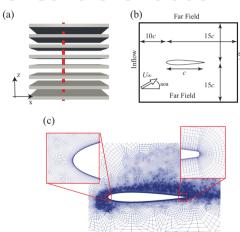
NACA0012 Aerodynamics coefficients

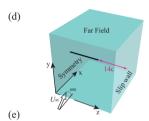


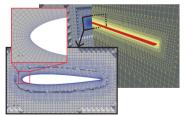




FSI of Cantilever blade





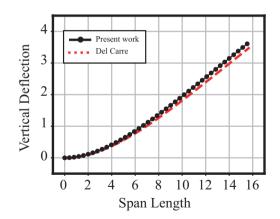






Cantilever Blade: Equilibrium before stall

- $Re = 1.56 \times 10^5$
- Angle of attack $\alpha_0 = 4^{\circ}$
- $g = 9.754 \, m/s^2$
- Del Care: beam + UVLM
- Number of Strips N_s = 4, 8, 12 tested. N_s = 8 selected

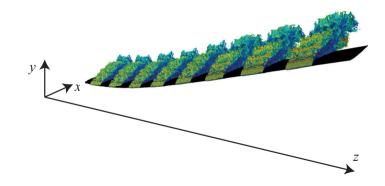






Cantilever Blade at stall

- $Re = 1.56 \times 10^5$
- $\alpha_0 = 16^{\circ}$
- $g = 9.754 \, m/s^2$
- 8 Strips
- Iso-contour *Q* = 10
- Color $|\mathbf{V}^*| = [0-2]$

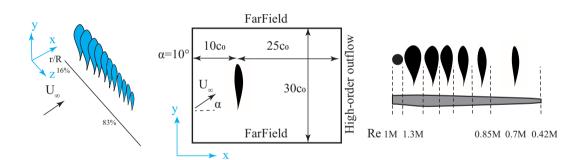






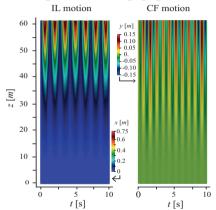


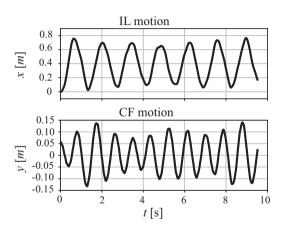
VIV of NREL5MW blade in parked condition

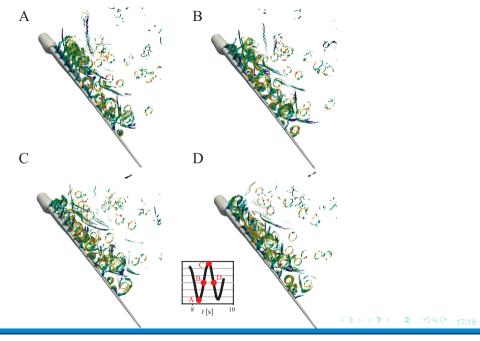




NREL5MW Blade VIV











Conclusion

- Two FSI solvers are developed for for highly-deformable slender structure
- Each targeted specific range of application
- Both needs high performance computations
- Comparison on the aerodynamic coefficients are presented
- Deformation of cantilevered beam and VIV of NREL Blade is presented for Nektar++/SHARPy Solver
- It is proven based on our experiment that coupling between OpenFoam and CalculiX using Precise is not straight forward
- CalculiX poses a HPC bottle neck in computations

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High Performance Computing in Wind Energy (https://www.hpcwe-project.eu/)

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