

Spline-based helicopter movement filter for sea ice surface topography estimation from airborne laser range finder

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Abstract

The surface topography of sea ice is important for several crucial climate processes, such as atmosphere-ice interaction, sea ice drift, snow redistribution and habitat conditions. A method, derived from the widely used three-steps Hibler filter, is suggested to gain quantitative surface topography information from airborne surveys over sea ice, when laser range finder are used. The method was tested and validated for several segments of a helicopter flight over Arctic sea ice. It was compared to a DTM generated from stereo-photogrammetry data collected simultaneously and independently from the laser data. The estimated ice surface profiles matched the reference DTM with an average absolute bias of 0.12 m.

Keywords: Sea ice, surface topography, laser range finder, airborne

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1. Introduction

Arctic sea ice processes play an important role in climate dynamics, local environment, shipping and tourism [1, 2]. In particular, the sea ice surface topography affects the atmosphere-ice-ocean interaction [3, 4], ice drift, snow redistribution [5, 6] and energy transfer, through the influence of summer melt [7, 8].

The use of laser range finders (LRF) or scanners has now well proven its efficiency in characterising the morphology of the sea ice surfaces [9, 10, 11].

However, in some applications like for example helicopter-borne electromagnetic sounding of sea ice (HEM) [12], LRF is used in the first place as an auxiliary setup only, not integrated with the GNSS and/or inertial navigation system unit. Then, the vertical motion of the aircraft blends in the recorded signal. With the amplitude of the aircraft motion being one to two order of magnitude higher than variation of the ice surface topography, the filtering procedure represents a crucial step in the processing of such data.

The three steps filter proposed by [13] has been used widely to estimate and remove the aircraft motion from the recorded signal. Although quite simple to implement and very efficient on short flight segments (<1 km long), this procedure show its limits when processing longer flights segments, in particular in the case of helicopter-borne surveys, because they potentially can exhibit higher variability in altitude over shorter distances than planes. The discontinuous nature of the procedure can also introduce distortions in the resulting signal.

Inspired by the procedure developed by [13], we here propose a spline-based helicopter motion filter to estimate sea ice surface topography from

26 LRF. We applied and tested the method on a dataset collected over Arctic
27 sea ice and compared it to a digital terrain model (DTM), created from
28 independent, simultaneously collected data.

29 **2. Helicopter movement filter**

30 The principle of the Hibler filter [13] for deriving ice surface topography
31 along flight lines relies on the assumption that the LRF signals from an
32 airborne platform comprise a relatively easy to disentangle superposition of
33 returns at a higher frequency, originating from the ice surface signal, and
34 lower frequency, originating from the aircraft motion. This assumption may
35 fit the reality quite well in the case of an airplane, with a fairly constant flying
36 speed. However, a helicopter can exhibit a much more complex motion profile
37 with the potential of more abrupt movements (although pilots usually avoid
38 this). In particular, helicopters have the possibility of changing altitude at
39 very low horizontal speed, resulting in effects of vertical helicopter motion to
40 leak into the high frequencies part of the altimeter series.

41 We here tested an empirical approach for the filtering of the helicopter
42 movement, based on the combination of the frequency analysis of the signal
43 and the estimation of the envelope of the signal. The proposed approach
44 is based on the assumption of the local maxima in the recorded raw laser
45 range profile, primarily associated with cracks and leads, which are common
46 features in the sea ice (Fig.1). These features can serve as reference points
47 creating a baseline to estimate the vertical motion of the helicopter, relative
48 to the sea level. It must be noted that depressions in the ice pack would
49 also appear as local minima in surface topography without necessary corre-

50 sponding to the sea level. A careful inspection of the recorded signal along
 51 with flight notes can help to connect visually observed cracks and leads to
 52 the respective altimeter data.

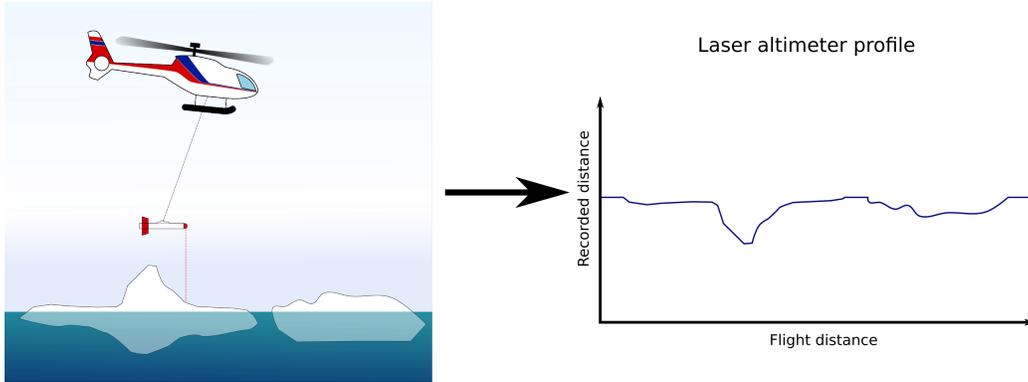


Figure 1: Schematic representation of a flight profile as recorded from the LRF of the HEM under ideal conditions (constant elevation relative to water surface). Lower distances for an idealistic level flight case correspond to ridges while higher distances correspond to cracks and open water.

53 Hence, we aim to recover the true motion of a helicopter relative to the lo-
 54 cal ocean surface using the estimation of a curve passing the largest recorded
 55 distance along the transect. Given that pilots usually steer helicopters in a
 56 way that they do not stop abruptly in horizontal flying direction, one can
 57 expect smooth transitions in the vertical motion of the aircraft. Hence, we
 58 can assume the helicopter motion to be continuous and fully differentiable
 59 along a flight track. So, the elevation data for the transitions can be approxi-
 60 mated by a polynomial function and estimated through a spline interpolation
 61 between the selected reference points.

62 For calculating the vertical motion profile along the flight distance, a
 63 two-step procedure is proposed, 1) filtering the raw LRF profile to isolate in

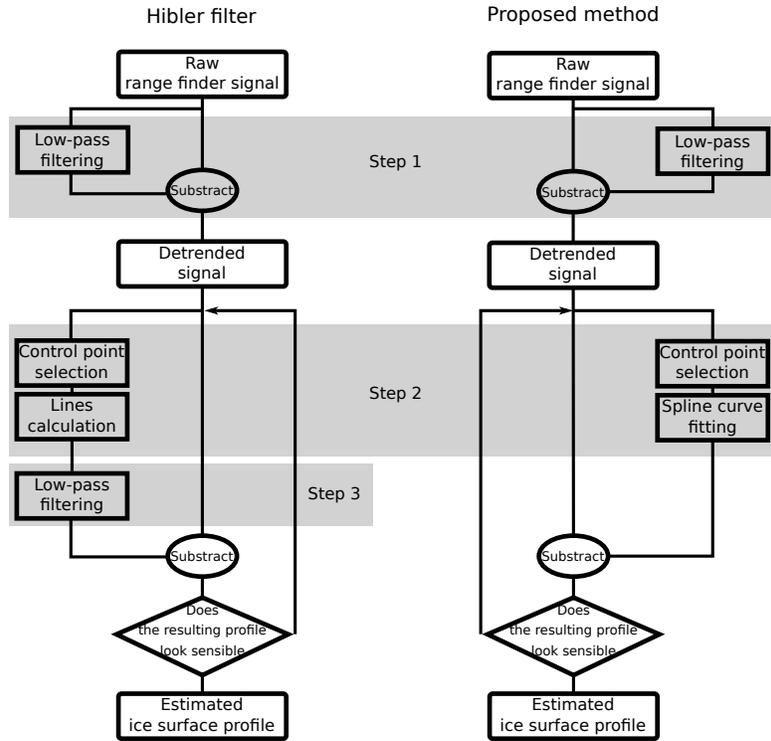


Figure 2: Filter process flowchart, from raw signal to the estimated ice surface profile. The left flowchart presents the Hibler filter while the right flowchart presents the proposed method.

64 the first approximation the vertical motion of a helicopter 2) curve fitting
 65 on reference point, similar in the initial approach to the Hibler filter (Fig.2).
 66 This proposed process significantly differs from the Hibler filter on the second
 67 step, directly estimating a curve instead of straight lines, and thus eliminate
 68 Hibler filter need for a low-pass filter at the third step to smooth the curve.

69 The first step actually consists of a low-pass filter to estimate a first es-
 70 timation of the vertical profile of the helicopter motion and subtract it from
 71 the raw signal (Fig.2). This approach simplifies the adjustment of the filter
 72 cut-off wavelength by visual interpretation. At this step a zero phase Butter-

73 worth filter [14, 15] with a cut-off wavelength of 200 m is applied. A typical
74 HEM flight is conducted with a speed of 30-40 m/sec; 200 m distance on the
75 ground corresponds roughly to 5 seconds flying time, which could be consid-
76 ered a typical timescale for a short range aerodynamic turbulence to disturb
77 the helicopter and the pilot reaction time to compensate for it. Depending on
78 flight conditions and pattern, applicability of this cut-off wavelength should
79 be assessed for each flight or even flight section, according to the main ver-
80 tical pattern of the recorded signal. For example, less calm flying conditions
81 may require a shorter cut-off wavelength to capture the more rapid changes
82 in elevation of the aircraft. This first step ensures a proper selection of the
83 reference points and allows for a better estimation of the helicopter motion
84 component remaining in the signal, such as rapid altitude losses or altitude
85 gains. The effect of the tuning of this cut-off wavelength will be further
86 discussed in section 5.

87 On step 2, a spline function is fitted to a set of reference points to further
88 generate a surface topography profile along the helicopter flight path. Refer-
89 ence points are automatically selected from the high-pass filtered raw LRF
90 profile by identifying the local maxima on a sliding window of a fixed width.
91 The number of selected reference points depends on this window length, so
92 the window length has to be adjusted according to the recorded signal and the
93 field notes (taken by the instrument operator during the flight) to guarantee
94 the best results. Here, from visual interpretation of the flight profile and the
95 plot of the selected points on the profile (Fig.3c), we selected a window of
96 175 m (corresponding to 500 measurement points), resulting in 66 reference
97 points selected. We will discuss the tuning of the parameters in more detail

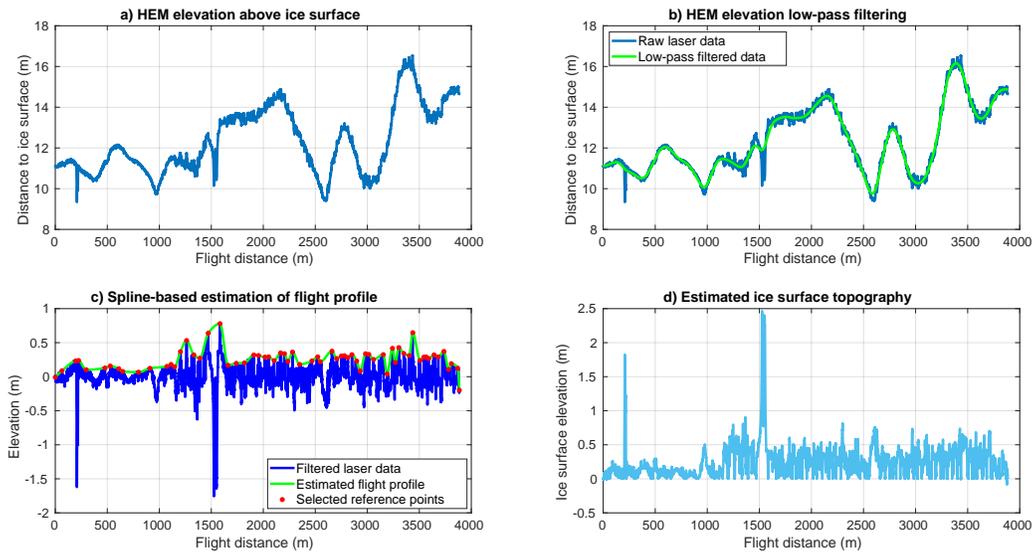


Figure 3: Example of the filtering workflow depicted in flowchart of Fig.2 applies on a 4 km flight section (see section 3 for details). The raw signal from LRF (blue line in a, b and c) is first detrended with a low-pass filter (green line in b) to help the selection of the reference points (red stars in c) for the calculation of the spline line (green line in c) representing the HEM vertical motion. The subtraction of the vertical motion leads to the estimation of the ice surface topography profile (purple line in d). The negative elevation at the end of the curve is the result of edge effects.

98 in the discussion section.

99 The superposition of the filtered signal generated in Step 1 and the spline
100 curve is assumed to be representative of the vertical motion of the helicopter
101 and can then be subtracted from the laser altimeter signal in order to recon-
102 struct the sea ice surface topography.

103 **3. Data**

104 For testing the described method, we applied it to data from a section of
105 an airborne survey in Fram Strait (Greenland Sea) over sea ice of the Norsk
106 Øer Ice Barrier (Fig.4), a large fast ice area east of Greenland [16, 17]. The
107 Fram Strait cruise is an annual interdisciplinary expedition led by the Nor-
108 wegian Polar Institute along a transect at 78°50' N latitude between Svalbard
109 and Greenland, consisting of research and monitoring within oceanographic,
110 sea ice physics, biochemistry and selected biology studies [18, 19, 20]. Some
111 years, it also comprises a helicopter for airborne survey. In 2016, the heli-
112 copter was equipped with both the HEM and the ICECam, a high resolution
113 camera setup with a capability to reconstruct the along-the-track surface
114 topography using photogrammetry. The combination of the two instruments
115 allowed simultaneous acquisition of the sea ice thickness and surface topog-
116 raphy. In this study, the ICECam data is used as a reference dataset to check
117 the validity of the proposed filtering method.

118 *3.1. Helicopter-borne electromagnetic ice thickness sensor (HEM)*

119 The ice thickness sensor makes use of the principles of electromagnetic
120 induction and the contrast of electrical conductivity of air and sea ice (low
121 conductivity) on one hand and sea water (high conductivity) on the other

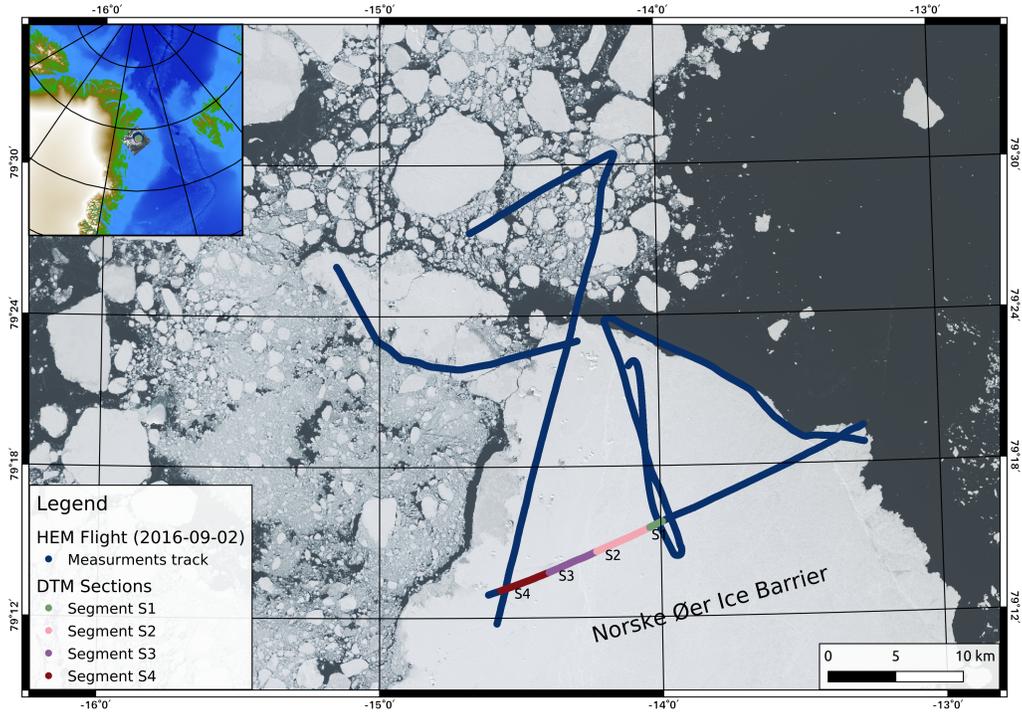


Figure 4: Map presenting the survey area, over sea ice of the Norske Øer Ice Barrier, east of Greenland. The marine blue represents the flight track with the HEM-measured total ice thickness (2 September 2016). Gaps in the track correspond to absence of measurements (either due to GPS signal loss or calibration of the instrument). The green, pink, purple and red sections represents the segments S1, S2, S3 and S4, respectively, where surface topography was reconstructed from ICEcam imagery using photogrammetric methods. The larger ice area marked "Norske Øer Ice Barrier" is static fast ice, an area of immobile sea ice anchored to several grounded icebergs, while the other ice in the image is mobile, drifting sea ice. Background: USGS/NASA Landsat 8 (7 September 2016)

122 hand [12]. Using a set of two coils (one transmitting and one receiving) allows
123 deriving the distance from HEM to the bottom of the sea ice. The total ice
124 thickness (ice plus snow) is calculated by subtracting the distance to the ice
125 or snow surface to the distance inferred to the sea water. During the flight,
126 the instrument is towed under the helicopter on a sling load of about 25
127 m (Fig.1), and has a typical altitude during measurements of around 15 m
128 above the ice surface.

129 The distance to the ice surface is measured using a Riegl LD90-3100HS
130 general purpose laser range finder mounted in the front of the instrument.
131 This LRF emits a beam of infrared light (905 nm wavelength) [21] which is
132 reflected by the top of the sea ice, or snow in the case of snow-covered sea ice.
133 With current settings the LRF measures a distance to the ice surface with
134 a sampling rate of 100 Hz and a typical accuracy of ± 3 cm [21, 22]. With a
135 typical flying speed of 30 to 40 m/s, the horizontal sampling distance along
136 the flight track varies from 30 to 40 cm.

137 Finally, position of the HEM during the flight is logged by an internal
138 GPS receiver.

139 *3.2. ICECam*

140 *3.2.1. Instrument*

141 The main components of the ICECam system are two downward looking
142 DSLR cameras, a GPS receiver, an inertial navigation system (INS) and a
143 LRF. The system is fitted into an aerodynamic pod mounted in the front of
144 the helicopter, with the GPS antenna mounted above the helicopter cockpit
145 [23, 24].

146 The shooting framerate is set at 1 image per second for each camera.

147 At the typical flying altitude during HEM surveys, around 40 m over the
148 ice surface for the helicopter, and the typical flying speed the camera pa-
149 rameters ensure a 50% to 70% overlap between the successive images with a
150 footprint of approximately 60 m by 40 m, at an altitude of 40 m over the ice
151 surface. This setup enables the reconstruction of the ice surface DTM, using
152 photogrammetry techniques.

153 *3.2.2. Processing ICECam data and generating the DTM*

154 To increase the accuracy of the positioning of the photos for the generation
155 of the DTM, the raw GPS and INS data is post-processed using the Precise
156 Point Positioning (PPP) technique. This is achieved with the commercial
157 software package TerraPos2 by TerraTec AS (Bergen, Norway). The resulting
158 elevation is given relative to the ellipsoid [25, 26]. Since tidal and atmospheric
159 effects can significantly affect the sea level, the ellipsoid-based positioning
160 data is adjusted using corrected for pitch and roll LRF data. The typical
161 offset between ellipsoid and ocean surface in the study area for the segments
162 of the flight track with photogrammetrically reconstructed DTM is about
163 25 m. For simplicity, we will further call the the post-processed positioning
164 data "GPS".

165 In order to build the DTM, the images are corrected for lens distortion
166 and vignetting. More detail on the ICECam system, the processing and its
167 accuracy can be found in [24]. Air refraction and Earth curvature are not
168 taken into account due to the low altitude the surveys are carried out at.
169 The reconstruction of the DTM is achieved using the commercial software
170 package Socet GXP from BAE-Systems. On test campaigns, the DTM has
171 proven to yield an RMS error of about 0.04 m and a bias of 0.03 m for the ice

172 freeboard, the distance between ice surface and water level, in comparison to
173 field measurements.

174 Since the processing and construction of the DTM is very time consuming
175 and computer-intensive, only short sections have been processed. In this
176 study, we are using four sections, called section S1 to S4 (Fig.4).

177 4. Results

178 In this section we show vertical motion and ice surface topography re-
179 construction results for the selected flight, and compare them with the data
180 from the ICECam. The HEM and ICECam datasets represent independent
181 observations, since no communication exists between the two instruments.

182 4.1. Helicopter vertical motion estimation

183 A flight altitude profile estimated using the proposed two-step approach
184 can be assumed to be representative of the helicopter motion. However, the
185 GPS profile from the ICECam is given relative to the ellipsoid and adjusted
186 to the local surface with the average distance measured with the own ICE-
187 Cam LRF, while the estimated profile is relative to the water surface. To
188 compare the two we will add the average difference between the two profiles
189 to the estimated curve. Fig.5a presents the comparison between the ICECam
190 GPS reference profile and the estimated profiles, for the four flight sections.
191 Fig.5b-e present the scatter plots of the HEM estimated profile against the
192 ICECam profile.

193 For all four segments, we notice that the two profiles do not overlap per-
194 fectly but the estimated profile looks reasonably similar the the GPS profile.

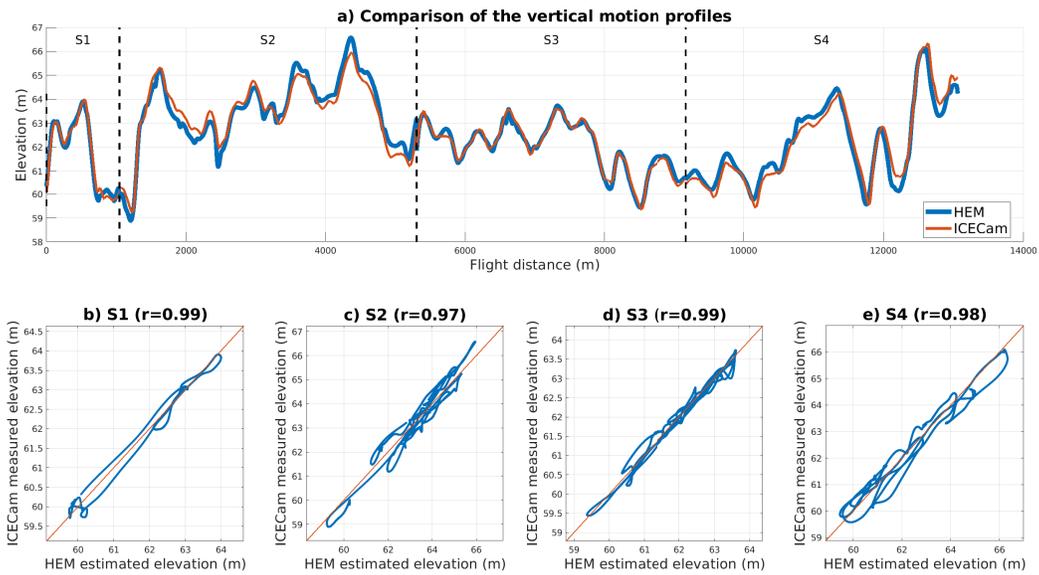


Figure 5: Comparison of the helicopter vertical motion estimation to the Precise Point Positioning flight altitude from the ICECam GPS+INS system. Plot a) represents the elevation data of the four segments (delimited by a vertical dashed line) for both the estimated motion (represented in blue) and the GPS profile (represented in red). The four plots b) c) d) and e) represent the scatter plots of the HEM estimated profile against the ICECam profile.

195 The correlation between the two profiles is systematically over 0.97 (Fig.5b-
196 e). The high correlation confirms the first visual interpretation of Fig.5. We
197 also calculated the mean absolute deviation and standard deviation of the
198 difference between the two profiles for each section. With an average mean
199 absolute deviation of 0.23 m (maximum of 0.33 m) and an average standard
200 deviation of 0.27 m (maximum of 0.37 m) we are within the acceptable posi-
201 tioning uncertainty for a PPP solution for a vertical coordinate [27].

202 Since the vertical motion of the helicopter has been proven to be estimated
203 within an acceptable precision, we can now in the next step use the HEM
204 LRF to estimate the sea ice surface topography.

205 *4.2. Sea ice surface topography reconstruction*

206 The DTM generated from the photogrammetric reconstruction of ICE-
207 Cam imagery is considered as the reference surface topography in this part
208 for comparison with the profile estimated from the LRF. However, we ac-
209 knowledge that a DTM calculated from photogrammetry may contain its
210 own uncertainties and possible biases, and it does not represent an ideal
211 ground truth.

212 To compare the surface profile estimated from the HEM LRF, the cor-
213 responding profile along the flight track was extracted from the DTM. Both
214 datasets are represented by point measurements, but the DTM has a much
215 higher spatial resolution than the HEM profile (5 cm for the DTM compared
216 to 35 cm, on average, for the altimeter), so we choose to extract the match-
217 ing profile by selecting the closest DTM point for each HEM point. The
218 comparison between the two surface profiles is presented in Fig.6a.

219 Since the GPS vertical profile from the ICECam and the estimated he-
220 licopter motion are not identical, the differences are expected to propagate
221 into the reconstructed ice surface. In the photogrammetrically reconstructed
222 DTM profile, we noticed negative values for some lower elevations while one
223 could expect the minimum values to match the sea level. It does not seem
224 random but more associated to a low frequency variability. This is most
225 likely the effect of uncertainties in the positioning and/or feature matching
226 processing in the photogrammetric solution, such as a lack of local z-control
227 points that causes biases in the vertical coordinate of the photogrammetric
228 reconstruction, and lies within the accuracy of the PPP solution. However,
229 to simplify the comparison with the HEM signal, we corrected this variability
230 by subtracting the lower envelope to the signal, a spline curve passing by the
231 lowest points in the signal, points assumed to correspond to the sea level.
232 For the calculation of the spline curve we followed the same procedure as for
233 the second step of our method.

234 Fig.6 demonstrates similarity between the ice surface topography esti-
235 mated from the HEM signal and the corrected ICECam DTM, apart from a
236 few exceptions in the DTM, probably corresponding to a crack in the ice or
237 melt pond (e.g. Fig.6b at 1500 m). This feature does not appear in the HEM
238 profile. The photos shows the edge of a meltpond, and few missing points in
239 the raw laser data. Infrared LRF signals are known to drop over meltpond
240 and open water. The reflected laser signal may then have been too weak for
241 the instrument to register the return.

242 Few more local deviation can also be noted for the example at about
243 2600 m (Fig.6c). This is most-likely the result of a local envelope-filtering

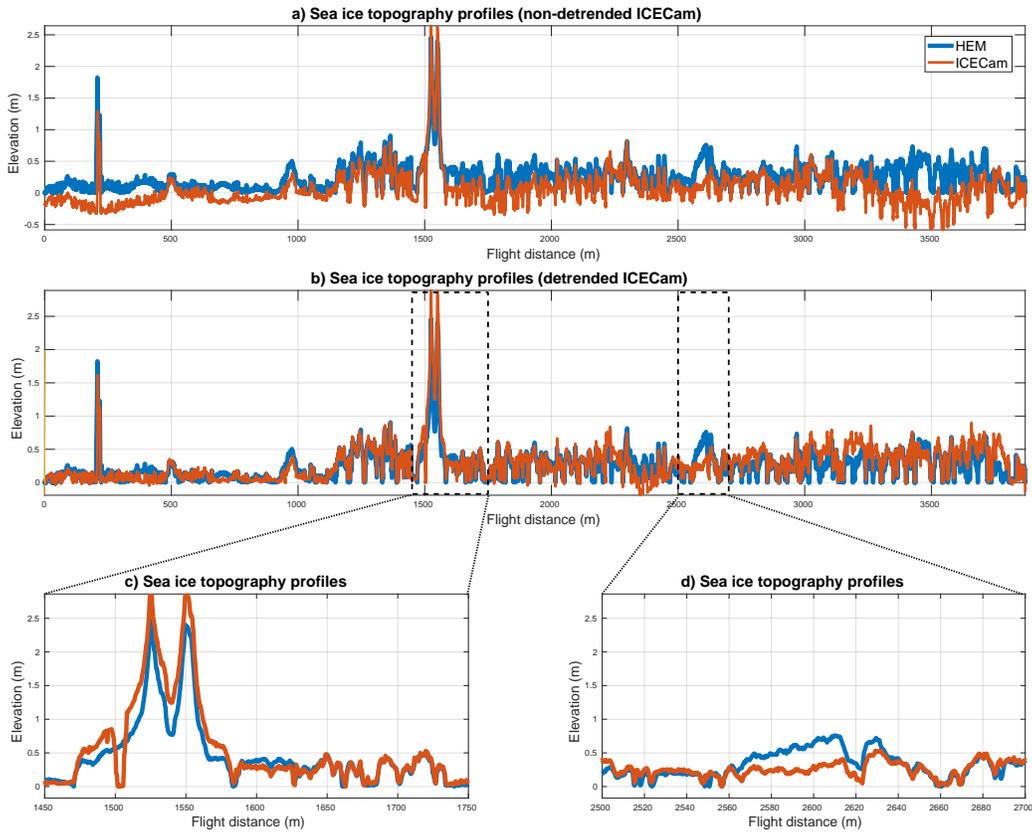


Figure 6: Comparison of the sea ice surface profiles, as estimated from the helicopter-borne electromagnetic sounder range finder (in blue) and the non-detrended ICECam (in red in plot a)) and detrended ICECam (in red in plot b)) for the flight segment S4. The rectangles in dashed-line in b) represents the bounding box of the zoomed version presented in c) and d).

244 artefact.

245 We applied the same procedure on all four flight segments (Fig.7). While
246 segments S3 and S4 provide satisfactory results with a correlation between
247 the two signals above 0.80, the two first segments seem less correlated, in
248 particular the second segment. Since all the segments were batch-processed
249 with the filtering parameters adjusted for the flight segment S4, a specific
250 tuning for each segment could improve these results. However, the main
251 features can still be recognised in all four segments. The standard deviation
252 is 0.18 m on average (for a mean absolute bias is of 0.12 m on average), which
253 is within acceptable margins (GPS PPP accuracy).

254 **5. Discussion**

255 *5.1. Comparison with the Hibler filter*

256 Since our proposed method is a modification of the Hibler filter, we com-
257 pared the output of the two procedures, using the method presented here,
258 and the Hibler filter. The first main difference is the number of parameters
259 to adjust to achieve a satisfactory filtering. The Hibler filter is a three steps
260 procedure, each steps having an adjustable parameter (Fig.2). By adding
261 one parameter, the Hibler filter has more degrees of freedom and hence more
262 ambiguous solutions. Moreover, being based on a succession of straight lines,
263 the two first steps of the process generate a discontinued signal. The low pass
264 filter of the third step is supposed to compensate for these discontinuities, but
265 may introduce non-negligible distortions due to a generally smooth nature
266 of the helicopter motion. The Hibler filter is then less adaptable to flights
267 with strong altitudinal changes, as it can be expected with a helicopter. This

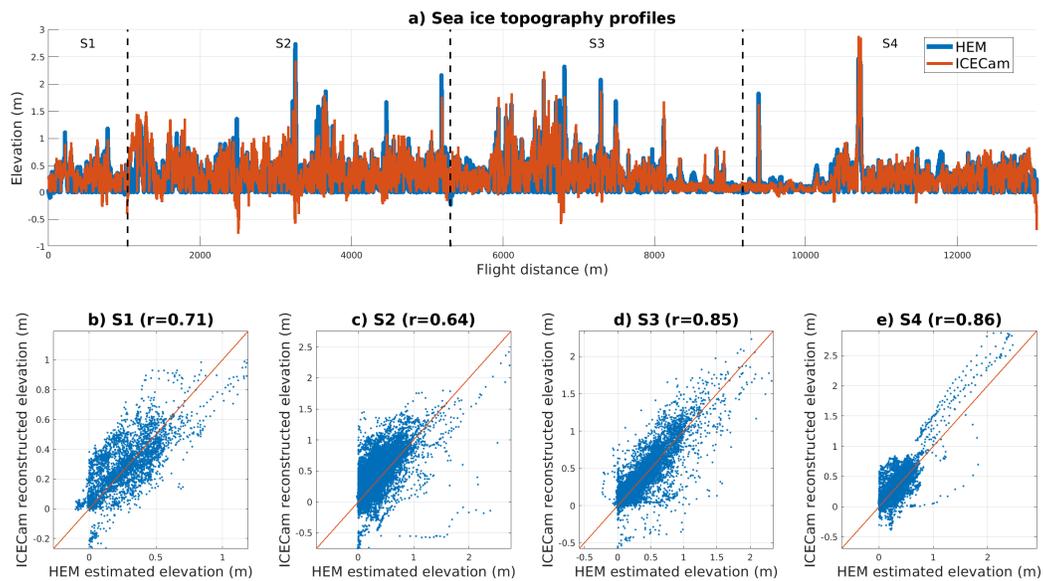


Figure 7: Comparison of the sea ice surface profiles, as estimated from the helicopter-borne electromagnetic sounder range finder using the Hibler filter (in blue) and the detrended ICECam (in red) for the four flight segments (a). The vertical lines delineate the different segments. The four plots b) c) d) and e) represent the scatter plots of the HEM estimated profile against the ICECam profile.

268 problem is however easier to solve when working on not too long flight sec-
269 tions, such as less than 1 km. The signal on such shorter sections presenting
270 in general less variability, the parameters can be tuned more specifically for
271 each flight segment.

272 For comparison with our filtering process, we applied the Hibler filter on
273 the same sections of the flight. Fig.8a presents the estimated helicopter mo-
274 tion as well as the estimated ice surface for section S4 using the Hibler filter.
275 As mentioned above, the filter has been difficult to tune properly. In par-
276 ticular, while choosing the right distance to select control points (Fig.2), we
277 had to look for balance between strong distortions (with longer lines on step
278 2) and the underestimation of some topography (with shorter lines on step
279 2). The second option seemed to provide the most realistic ice topography.

280 The first 1000 m looks properly filtered, and the resulting ice surface pro-
281 file presents very few negative values. However, when the altitude profile
282 starts exhibiting mainly ridges with little to no level ice, the Hibler filter
283 performance drops significantly. This can also be seen on the estimated ice
284 surface on Fig.8b. After the first 1000 m, the number of negative values in-
285 creases significantly. The correlation (0.73) appears weaker than with our
286 filter and both the bias (± 0.16 m) and standard deviation (0.19 m) are also
287 higher than with our filter. We can note these results do not discard the ap-
288 plicability of the Hibler filter but confirm our method provides improvements
289 in the quality of the final product.

290 As previously, we repeated the comparison process over the four segments
291 of flights (Fig.9). The results appear similar for the four flight segments.
292 The average correlation (0.66) is lower than with our filter and both the bias

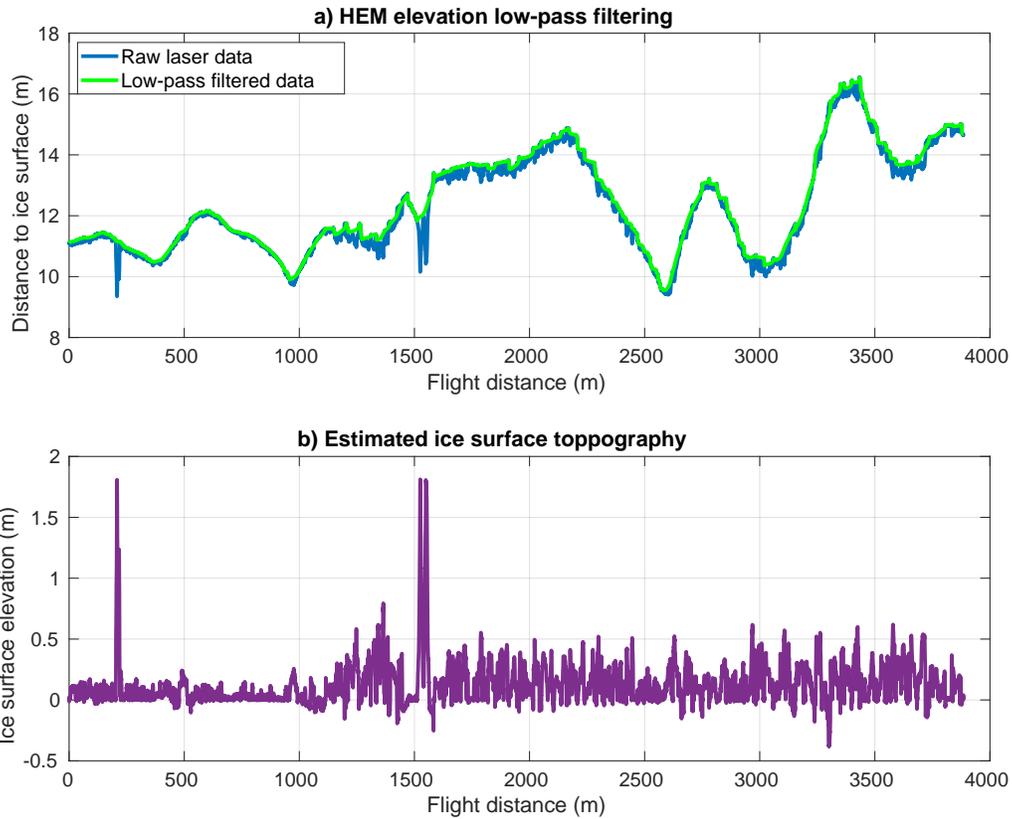


Figure 8: Filtering of the section S4 using the Hibler filter. a) Raw signal from the LRF and the helicopter motion estimated using the Hibler filter. b) Estimated ice surface topography.

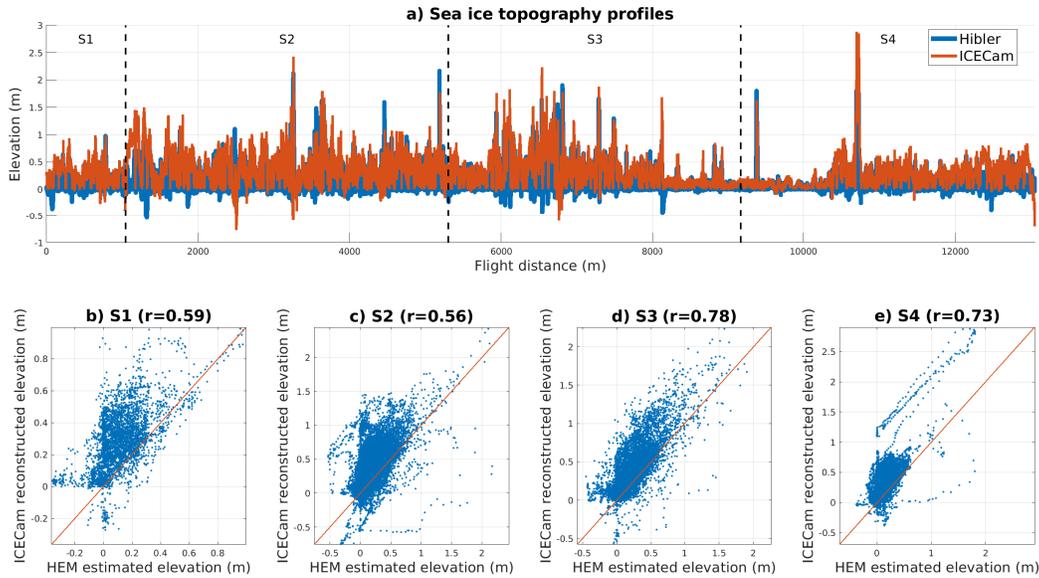


Figure 9: Comparison of the sea ice surface profiles, as estimated from the HEM LRF using the Hibler filter (blue) and the detrended ICECam (red) for the four flight segments. The vertical lines delineates the different segments.

293 (± 0.19 m) and the standard deviation (0.20 m) are higher.

294 We calculated the distribution of the estimated ice surface elevation for
 295 the ICECam data and the two filtering method, for each segments of flights
 296 (Fig.10). Our filtering method appears to generally match the ICECam
 297 distribution for the segments S3 and S4. Our method seems to over-estimate
 298 the amount of zero-elevation data on both segments S1 and S2. It also
 299 misses a second mode around 0.25 m on segment S1 while presenting one
 300 not present in the ICECam data on S2. The Hilber method consistently
 301 over-represents the lower elevation while under-estimating everything above
 302 0.25 m. This is consistent with the choice we had to make when adjusting
 303 the filter parameters, in particular the line length in the second step.

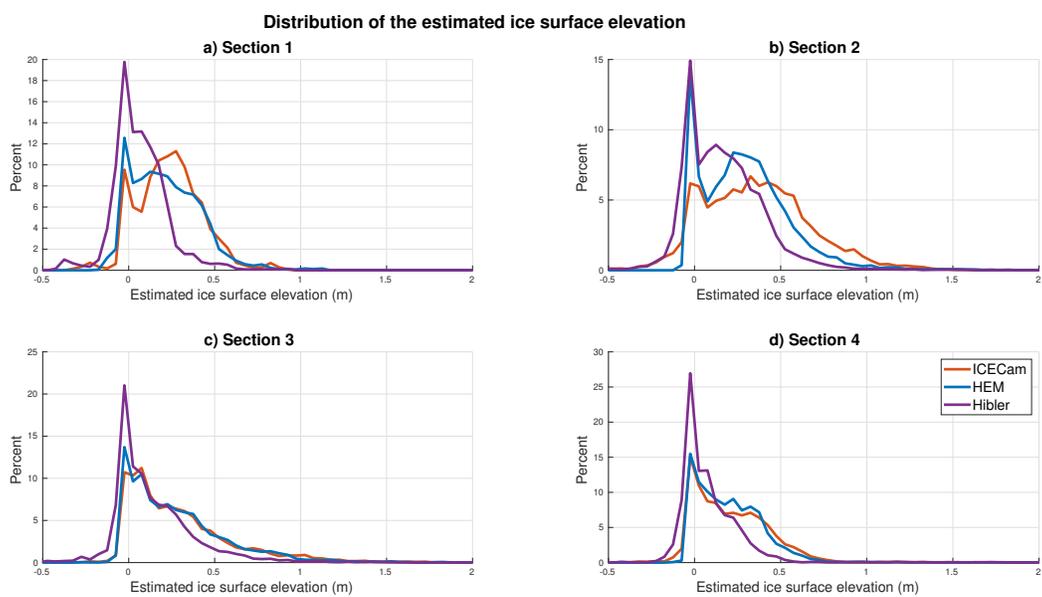


Figure 10: Comparison of the sea ice surface elevation distribution, as estimated from the detrended ICECam (red), our filtering method (blue) and the Hibler filter (purple) and for the four flight segments.

	ICECam		Proposed filter				Hibler			
	mean	std	Elevation		Bias		Elevation		Bias	
	mean	std	mean	std	mean	std	mean	std	mean	std
S2	0.29	0.19	0.26	0.20	0.21	0.27	0.12	0.16	0.24	0.24
S2	0.41	0.32	0.34	0.27	0.32	0.42	0.22	0.23	0.35	0.41
S3	0.34	0.31	0.33	0.30	0.25	0.37	0.19	0.24	0.26	0.36
S4	0.27	0.28	0.26	0.23	0.19	0.28	0.13	0.18	0.21	0.29

Table 1: Comparison of the ice surface elevation statistics as estimated by the ICECam and the HEM LRF using both our proposed filtering method and Hibler filter.

304 Overall, the ice surface elevation statistics for all four flight segments, as
305 well as bias, appears in favour of our method (Tab.1).

306 5.2. Adjusting the filtering parameters

307 The main challenge we experienced in this procedure is to manage to
308 adjust the Hibler’s filter parameter to reach acceptable results. As mentioned
309 previously, the extra degree of freedom leads to a higher sensitivity of the
310 method at the expense of labor intensity. Therefore we had to spend a
311 large amount of time to adjust the parameters for the results presented here.
312 However, the method proposed in this study also shows to be sensitive to the
313 filtering parameters.

314 In our method, the first step (low-pass filter, Fig.2) of the filter plays an
315 important role. If the main component of the helicopter movement is not
316 filtered, the amplitudes of the variations are too strong. The spline curve
317 then tends to oscillate and results in strong distortion of the final signal. But
318 the overall process remains less sensitive to the cut-off frequency at this step.

319 Our empirical estimation of this parameter allowed us to remove most of the
320 helicopter motion and no readjustment was required.

321 The sliding window length to select the control points is the most crucial
322 parameter. If we select too many points (with a shorter sliding window),
323 the spline curve will tend to follow small variations in the laser signal and
324 underestimate the height of pronounced topographic features such as ice
325 ridges. On the other hand, selecting a window too large will result in too
326 few control points and prevent the curve to match properly the laser signal.
327 This can lead to distortions and negative values of the ice surface elevation.
328 Optimal choices for these processing variables depends to a high degree on
329 the nature of the sea ice in a region, including the ice concentration, and the
330 flight conditions and quality of the raw data.

331 The results from the processing of the airborne data are further on com-
332 pared to field notes and on-flight imagery if available, in particular to check
333 if the ridges and the leads are correctly represented in the final ice surface
334 topography. If the results do not match the field notes, the length of the
335 sliding window may be adjusted accordingly to reach the expected results.
336 The adjustment of the first cut-off frequency should only be considered if the
337 intermediate signal still presents significant low-frequency amplitudes.

338 To analyse the impact of the filtering parameters, we applied the proposed
339 method on each of the four flight segments using a filter cut-off wavelength
340 (Step 1, Fig.2) and sliding window length for the control points selection
341 (Step 2, Fig.2) ranging from 10 m to 1500 m (with a 10 m step). Each re-
342 sulting estimated ice surface is compared to the ICECam DTM. The corre-
343 sponding absolute bias and standard deviation are calculated and plotted in

344 a matrix (Fig.11). Contour lines are added on each plot panel to delineate
345 0.15 m and 0.25 m thresholds. For each of the four flight segments, the lower
346 values of absolute bias are mainly found in a band corresponding to a filter
347 cut-off wavelength ranging from 100 m to 400 m. However, the sliding win-
348 dow length seems to play only little role on the bias. This can be explained
349 by the magnitude of variation each step of the filter is working. The first
350 step aims to remove a signal having an amplitude of several meters, while
351 the second step filters amplitude of less than a meter. In particular, we can
352 notice that our 200 m cut-off wavelength for the first step appears suitable
353 for all the flight segments while the window length should be adjusted for
354 each segments. The window length parameter seems to play a more impor-
355 tant role in the standard deviation. Here again, a common 200 m cut-off
356 wavelength for the first filtering step provide good results for all the flight
357 segments while the window length need to be adjusted for each segments to
358 reach the best possible result (ranging from almost 700 m for S1 to 200 m for
359 S2).

360 *5.3. Laser range finder over open water*

361 The HEM LRF emit pulses at infrared wavelength. This wavelength is
362 poorly reflected by sea water, then the returned signal to the instrument
363 can be too weak to be registered. In such case the instrument records a
364 "missing value". It can result in fewer valid values recorded over leads and
365 melponds, when they are melted through. This can pose an issue for the
366 filtering procedure since it is based upon the assumption the lower values
367 in the laser signal represent the sea water level. In practice, the LRF often
368 manage to record few points, and the field notes can help identifying the leads

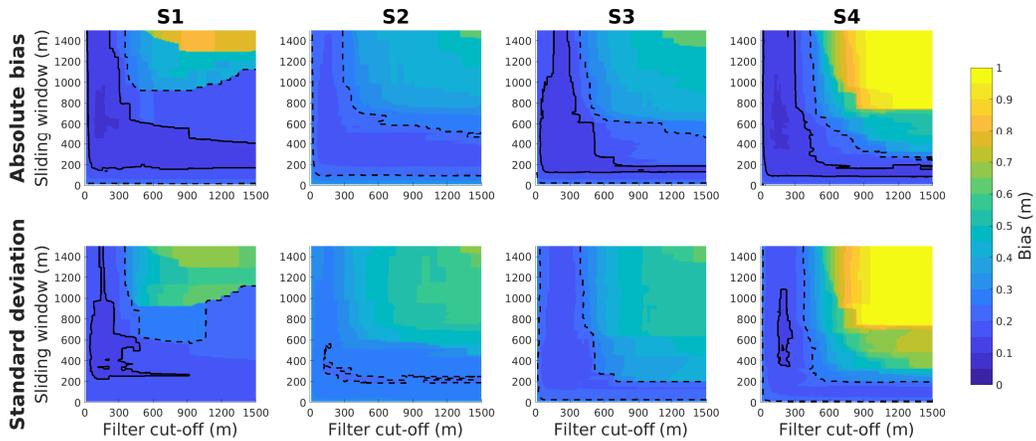


Figure 11: Absolute bias and standard deviation of the estimated ice surface topography transects, compared to the ICECam DTM, for filtering parameters ranging from 10 m to 1500 m for the four flight segments S1-4. The solid black contour line represents a 0.15 m threshold and the dashed black contour line a 0.25 m threshold.

369 and check which portion of the signal is missing. When only short distances
 370 are missing, an interpolation can then be used to compensate for the missing
 371 data. Longer missing segments are unlikely to happen since helicopter pilots
 372 usually avoid flying at low measurement altitude over open water.

373 In the eventuality of such long missing segments, in the order of a hundred
 374 meters or more, interpolation seems unreasonable and the signal should then
 375 be treated as two different flight segments.

376 *5.4. Transects without open water*

377 Since recording over open water is important to estimate the reference
 378 level for the helicopter motion, a flight happening over packed ice and vast
 379 sea ice floes without any openings also poses a problem. In such case, we
 380 have to assume the level ice to be parallel to the water surface and use it as a

381 reference. Although the relative changes in the topography of the ice surface
382 is preserved, which would be sufficient for applications such as, for example,
383 snow drift or radar back-scattering models, the overall heights above sea
384 level will be underestimated. In practice, open water and leads are usually
385 found around the ship from where helicopter-borne surveys start. By actively
386 choosing such area in a transect, a reference level can be recorded ensuring
387 better result from the method.

388 **6. Conclusion**

389 We investigated a spline-based helicopter motion filter to estimate the
390 sea ice surface topography from laser range finder. The filter is inspired of
391 the widely used three-steps Hibler filter [13]. Our proposed method shares
392 its first step with the Hibler filter, but differs at the second steps by using
393 a spline line instead of a succession of straight segments. This eliminates
394 the problem of discontinuities in the signal and the need for a third step to
395 smooth the discontinuities with a low-pass filter, and thus an extra tuning
396 parameter.

397 We tested our method on data collected during an NPI expedition to
398 Fram Strait in 2016 and compared it to a DTM generated from stereo-
399 photogrammetry data collected simultaneously and independently from the
400 HEM data. The stereo-photogrammetry system incorporates a GPS which
401 also gives us a reference for the helicopter vertical motion. In both cases,
402 helicopter vertical motion and ice surface topography, the mean absolute
403 bias of our model falls under the expected accuracy of GPS Precise Point
404 Positioning reprocessing.

405 Although the Hibler filter gave overall satisfactory results, our procedure
406 systematically resulted in a lower absolute bias. Furthermore, we experienced
407 the spline-based filter easier to tune, having less parameters to adjust (two
408 instead of three). This is an advantage when processing longer flight segments
409 (several kilometers) and helicopter-borne surveys, helicopter being able to
410 change altitude more abruptly than planes.

411 The method could only be tested on four flight segments, for a total
412 distance of 13 km, presenting overall similar ice conditions. As a next step,
413 a comparison using more diverse ice conditions could help to better identify
414 the limitation of such procedure, in particular in different sea ice conditions,
415 with a variety of ice concentration and sea ice types and features.

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424 ice in the Arctic Ocean, Technology and Governance" flagship.

425 The Matlab script of the filter function is available on Github: [https://](https://github.com/jnegrel/HeliFilter)
426 github.com/jnegrel/HeliFilter. The data used for this study is available
427 upon request.

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